

Active Multi-Modes Vibration Control of PZT Honeycomb Sandwich Plates

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ABSTRACT

The purpose of this project is to develop an active multiple mode vibration control of PZT honeycomb sandwich plate system. Active control in multiple modes of vibration of structures remains an engineering challenge issue. The research topics of this issue include the minimum numbers of sensors/actuators, the optimum locations of sensors/actuators, the development of simplified mathematic model for control design, the minimum control effort and the stability, robustness of the controller. The PZT adaptive remains that were developed at the Chung-Shan Institute of Science and Technology, which has been utilized in the area of structural vibration control and flutter suppression. The d31 type piezoelectric actuators evaluated in this study include PZT with inter digital electrodes (IDE) and Macro Fiber Composites (MFC). Interdigitated electrodes (IDEs) are used for poling and to direct the electric field along the axis with major actuation of PZT. The PZT with inter digital electrodes actuators achieve greater actuation energy density by exploiting the d31 effect versus the d31 actuation used in most monolithic piezoceramic materials. The analysis for the composite plate model with d31 type piezoelectric actuators conducted by the laminate plate theory accounts for the anisotropic characteristics of d31 type piezoelectric actuators/composite plate. This model will be used to develop the active controller in suppression structural vibration and flutter. The candidates of active controller include the velocity feedback control, the LQR method and the filter-x LMS method. The active control system designed for flutter suppression and vibration attenuation requires the equation of motion to be expressed in a linear time-invariant state-space form. The doublet-lattice method is utilized to compute unsteady aerodynamic forces, which are approximated as the transfer functions of the Laplace variable by the minimum-state method. The feedback control gains are obtained by solving coupled nonlinear matrix equations via numerical routines. This project will demonstrate the effectiveness of vibration reduction of the d31 type piezoelectric actuators/composite plate in the aspects of the theoretical predictions and experimental validations. Moreover, numerical simulation of flutter suppression of the d31 type piezoelectric actuators/composite plate will be presented in this study.

Keywords : macro fiber composites, vibration reduction, flutter control

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